

# Enabling Lunar Exploration:

## ESA's PNT Roadmap & User Benefits

**UN Cislunar Workshop**  
10 – 13 February 2026  
Vienna, Austria



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Richard Swinden, Yoann Audet,  
Floor Melman and Serena Molli  
European Space Agency



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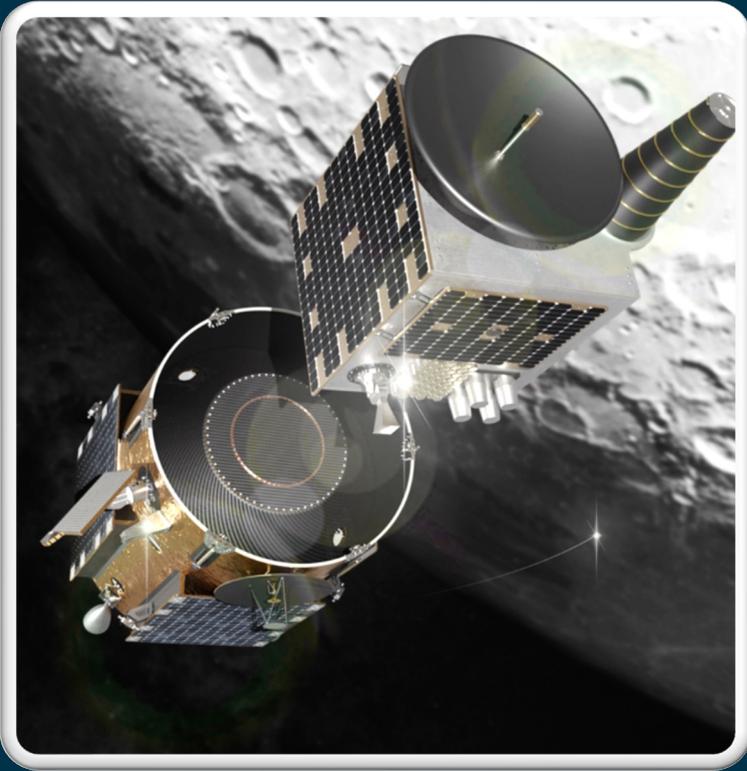


# ARTEMIS-2 upcoming launch with ESA's European Service Module in ORION

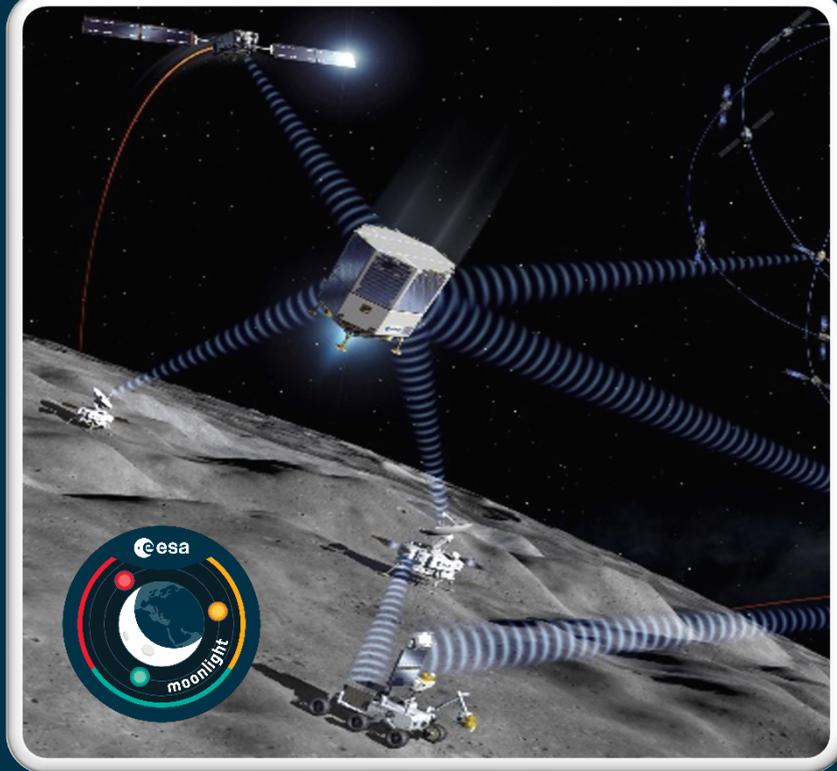


***First human return time to the Moon since 1972 !***

# ESA Roadmap For Lunar PNT Services



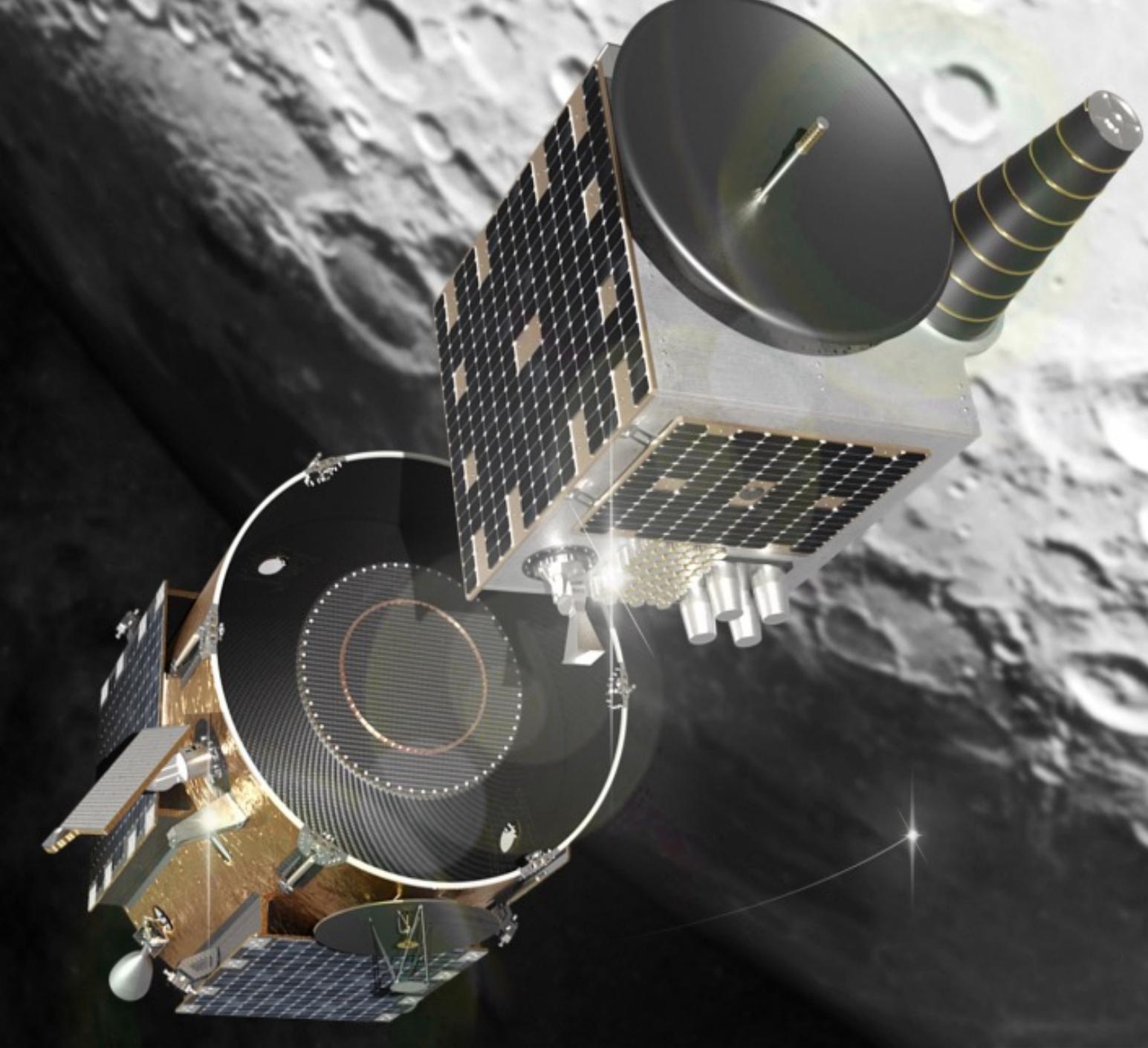
**STEP 1:  
LUNAR PATHFINDER**  
**(Launch 2027)**



**STEP 2:  
MOONLIGHT System**  
**(Initial: 2029; Final: 2031)**

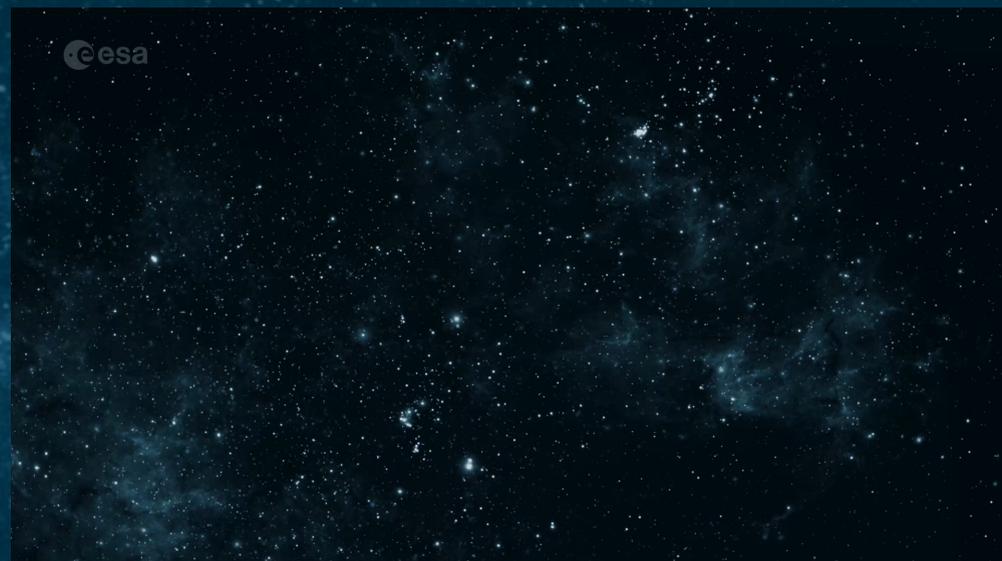


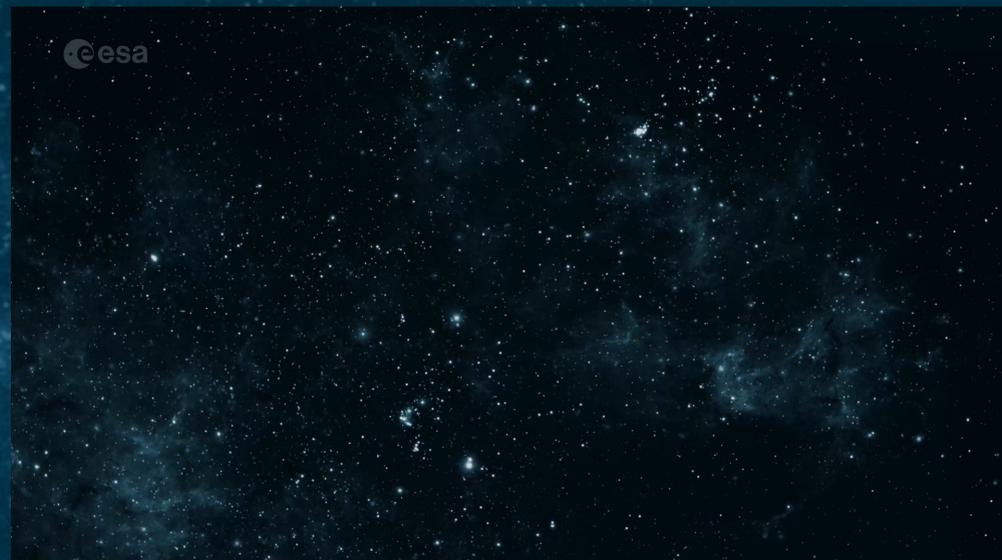
**STEP 3:  
NOVAMOON: Lunar-based  
PNT Differential Station**  
**(Launch target: 2029+)**



- **Commercial** Lunar Communication **Relay** Satellite (real time S-band services) (**Moonlight Step-1**)
- Public-Private Partnership (**PPP**) between **ESA** and Surrey Satellite Technology Limited (**SSTL- UK**)
- Will provide data relay operation for the **first US lander on the far side (LuSEE-Night)**
- **Launch:** Q1 2027
- FireFly's Blue Ghost CLPS-CS3
- **8.5 years lifetime**

**Will host a high-sensitive GNSS receiver on board**

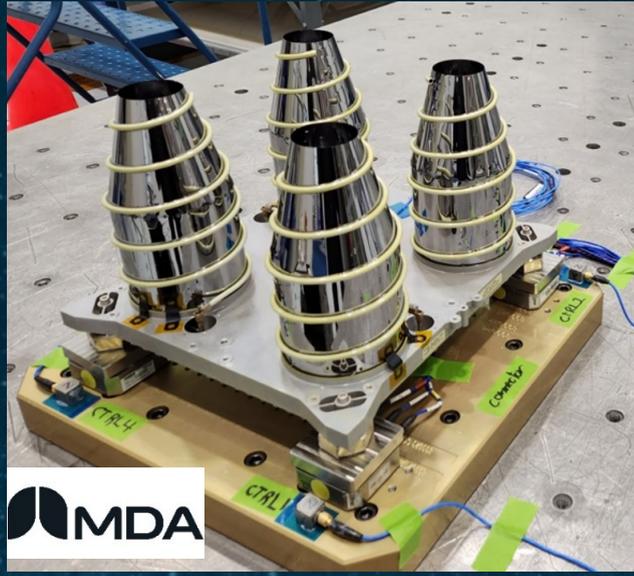




# Lunar Pathfinder GPS/ Galileo Reception



SpacePNT+



MDA



NASA

GNSS High-sensitive receiver Flight unit

GNSS High-gain Antenna Flight unit

Laser Retroreflector Flight unit (NASA)

Full demonstration of GPS/Galileo PNT on a Lunar orbiting satellite

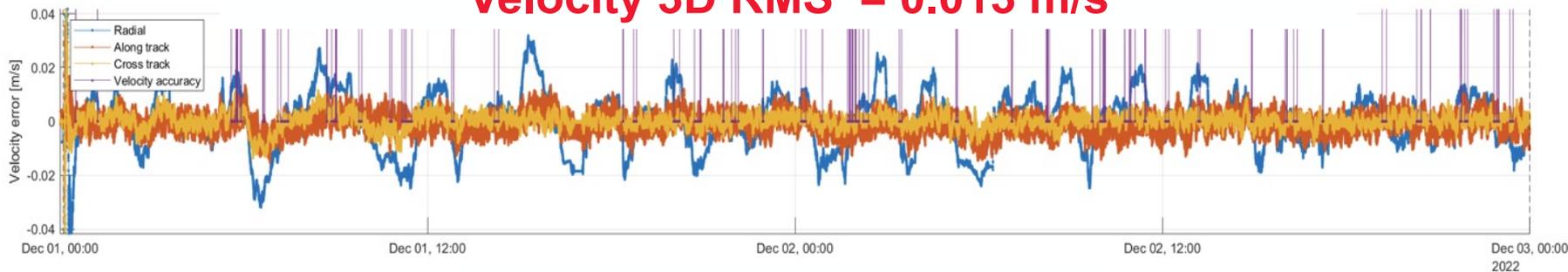
First time three ranging techniques (GNSS, Laser and X-band ranging) are used simultaneously on lunar orbit for orbit determination

# Lunar Pathfinder GNSS Rx end-to-end hardware in the loop updated results with delivered Flight GNSS Rx HW and on-board SW

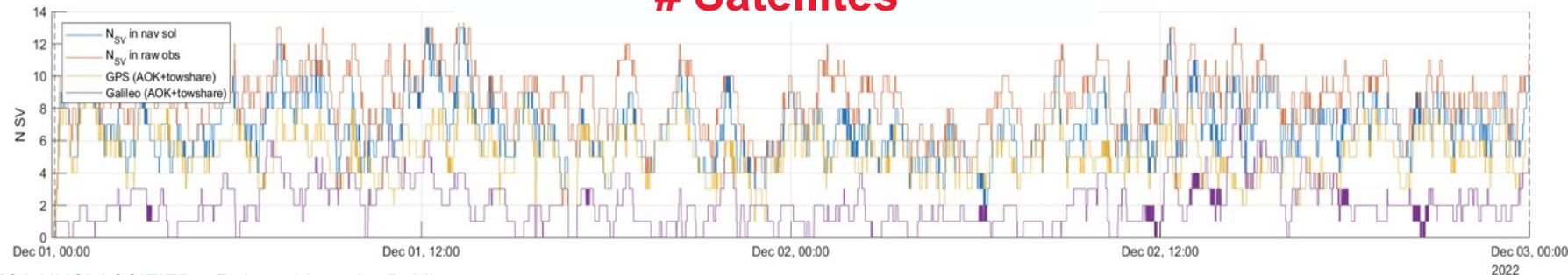
**Position Accuracy 3D RMS: 32.84 m**



**Velocity 3D RMS = 0.013 m/s**



**# Satellites**



Well below 100m 3D rms initial Requirement on lunar orbit!

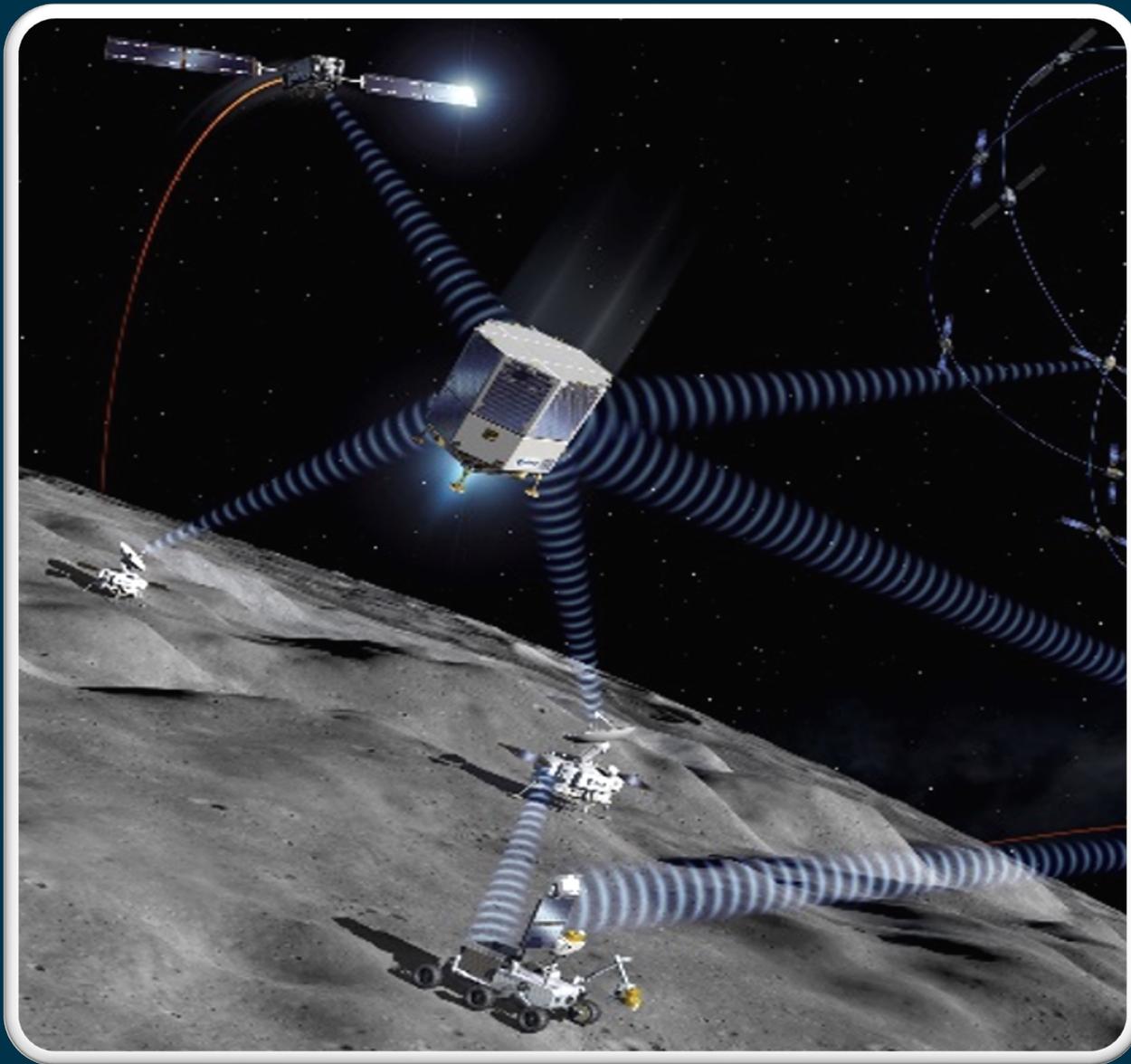
SpacePNT Lunar Pathfinder GNSS Receivers provides today the space receiver with the highest sensitivity

(with 15 dB tracking threshold and 18 dB acquisition threshold !)

# Lunar Pathfinder GNSS Experiment: User Benefits

- Following launch, **dedicated experimental windows** (~ 5 to 7 days with continuous Earth visibility) are planned, using also laser ranging stations **from Apache (USA), Grasse (France), Wetzell (Germany) and Matera (Italy)**.
- This should allow confirming **operational benefits of GNSS for lunar orbit determination and time transfer**
- **The 8.5 years nominal life-time of Lunar Pathfinder** will also allow:
  - Upgrades on GNSS Receiver Software for improvements based on experimentation return
  - Assess GNSS cislunar performance with future Galileo G2G and GPS IIIIF satellites
  - Collection long-term GNSS dataset in lunar orbit for international scientific analysis
  - To perform joint experiments with GNSS Receiver in orbit and lunar surface (with NovaMoon)





**STEP 2**

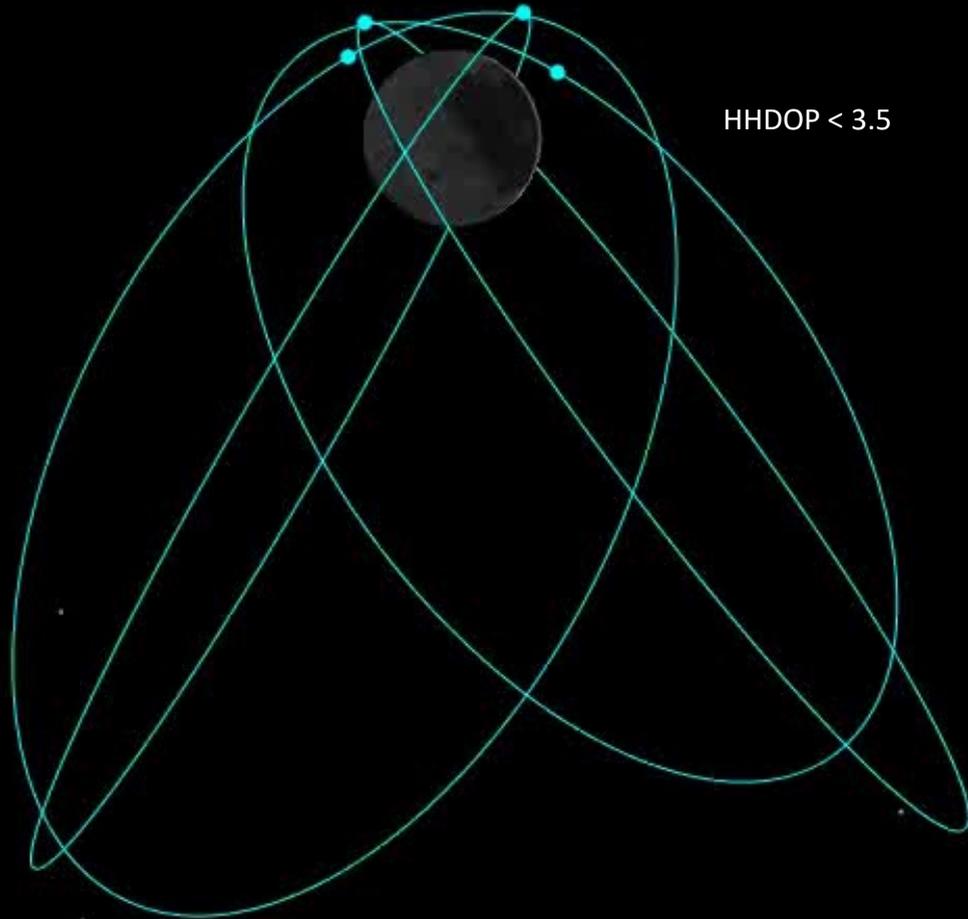
**MOONLIGHT System**

**IOC: 2029 & FOC: 2031**

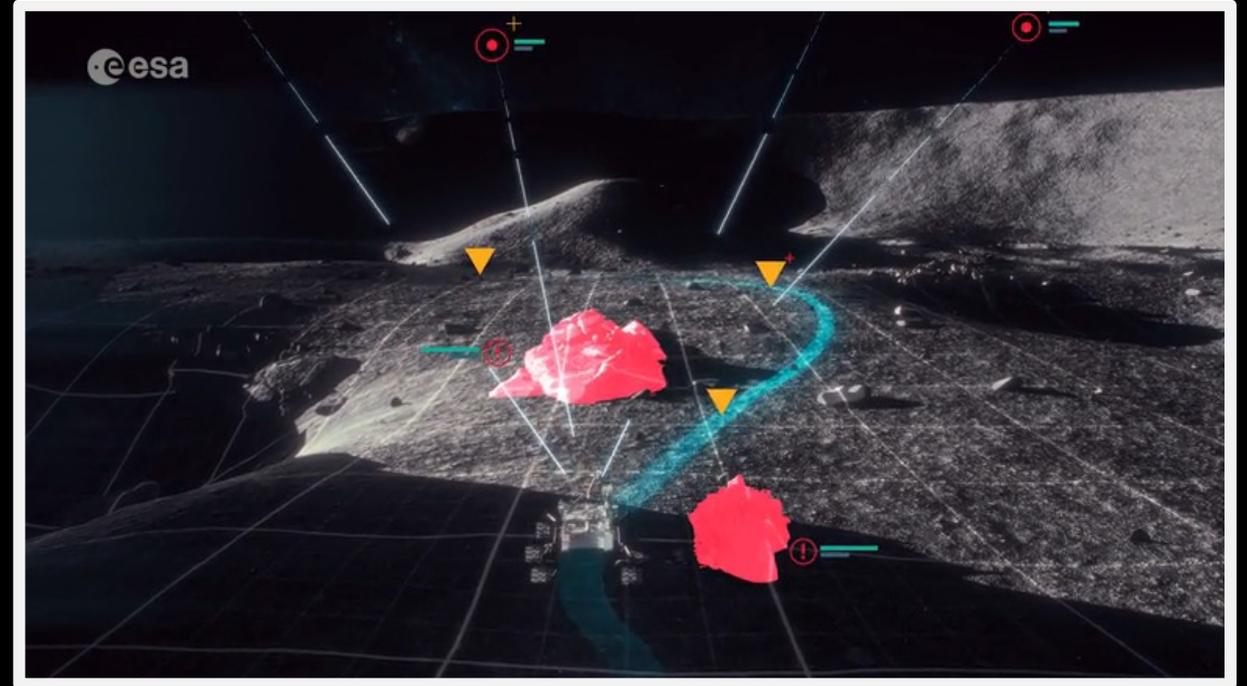




# Moonlight 4 NAV Satellites constellation

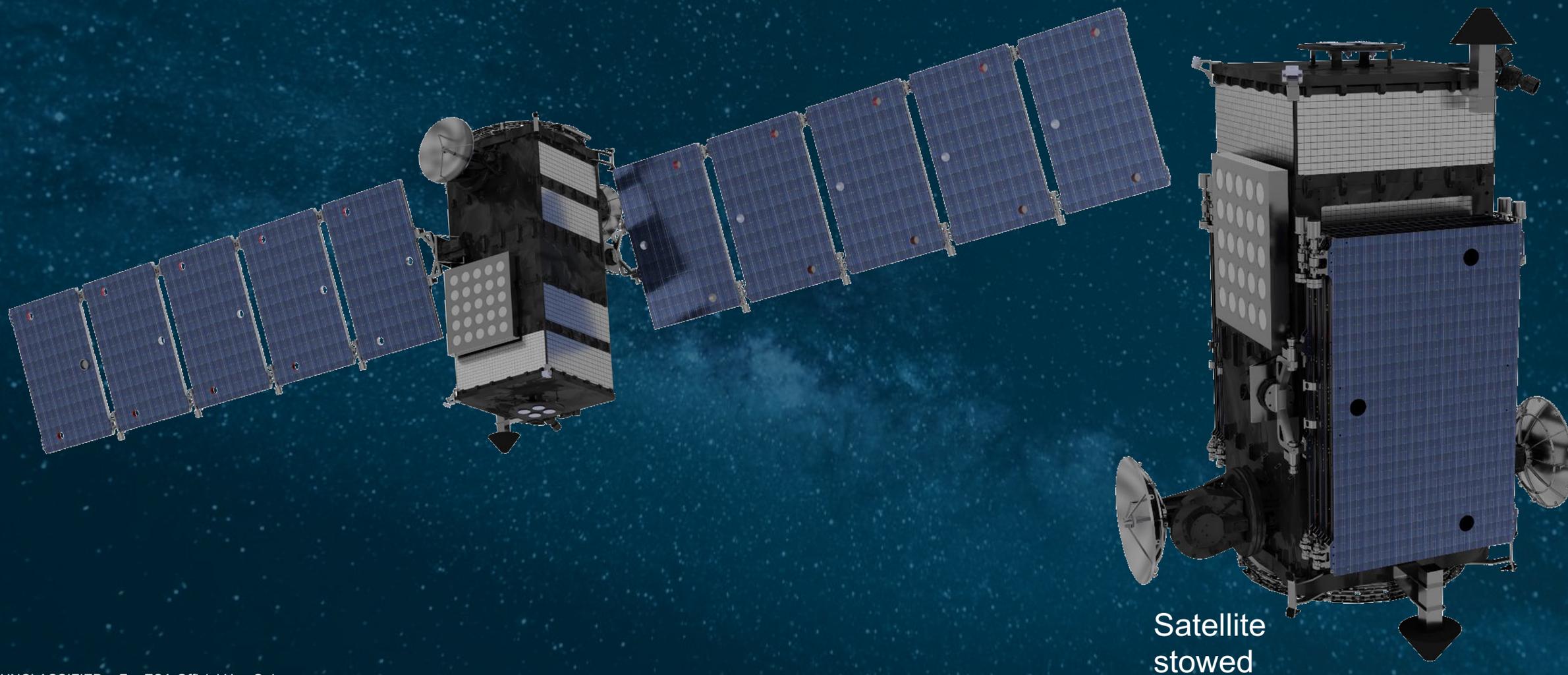


**24h ELFO ORBITS**

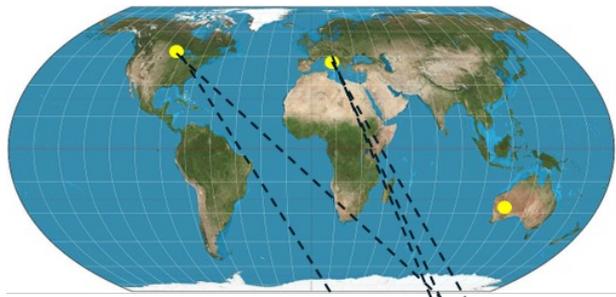


A minimum of  
15 hours of continuous PNT service at South  
Pole every 24 h

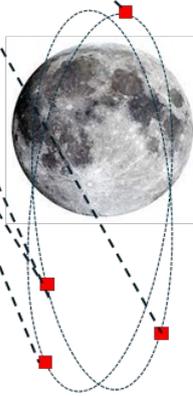
# Moonlight NAVSAT



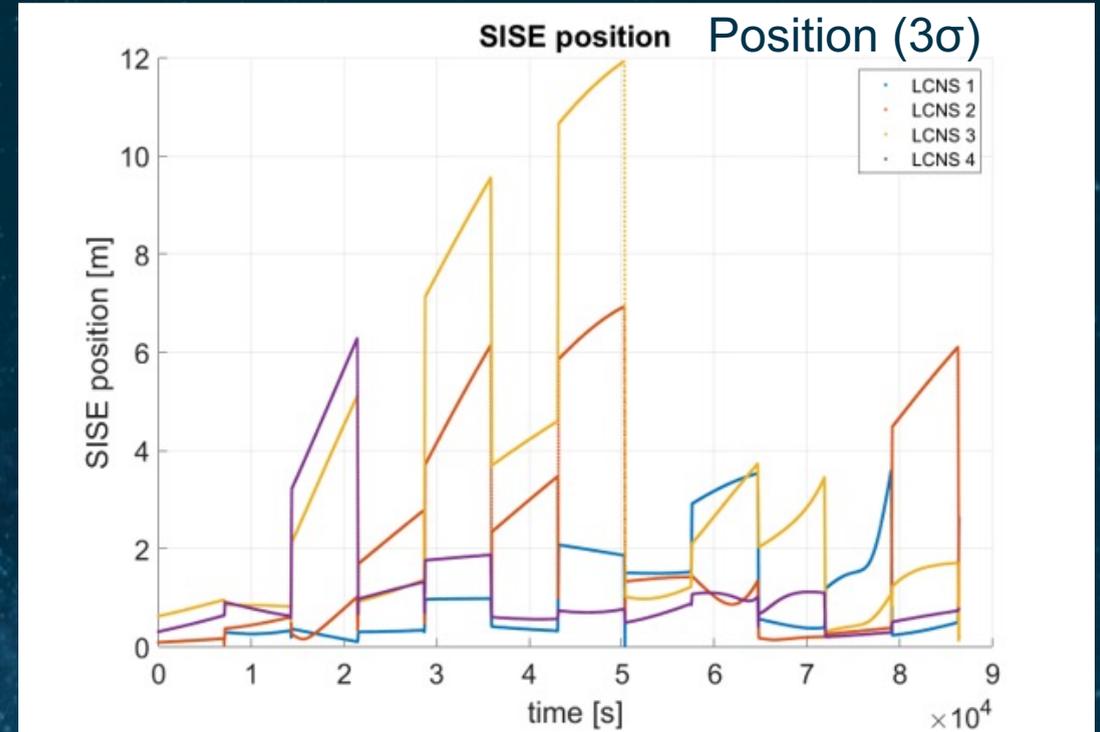
# Moonlight ODTS Concept: MSPA Technology



Multiple  
Spacecrafts per  
Aperture (MSPA)



$$SISE_{pos} = \sqrt{(x - \tilde{x})^2 + (y - \tilde{y})^2 + (z - \tilde{z})^2 + (ct - \tilde{ct})^2} < 10 \text{ m (95\%)} \\ \text{over any 24 hours}$$

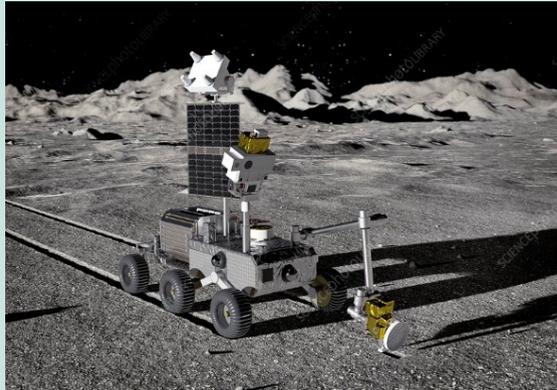


- Two-Way Time Transfer between NAV SATs and EGS and use of stable Atomic clocks on board
- Ranging and Doppler measurements between NAV SATs EGS
- Navigation message generated from EGS and uplinked to NAV sats

# Moonlight FOC LCNS High-level PNT Performances



## Surface Rover



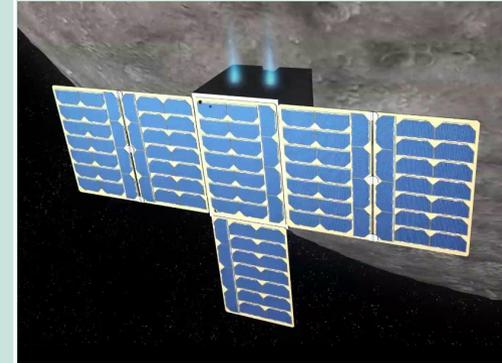
Real time  
**< 10 m (95%)**  
 Post-processing  
**< 3 m (95%)**

## Lunar Lander

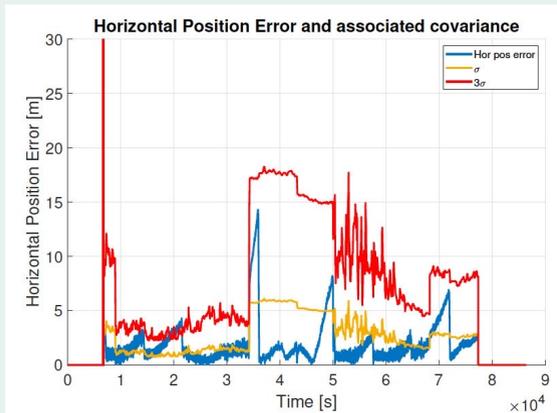


**< 50 m (95%)**  
 Landing accuracy

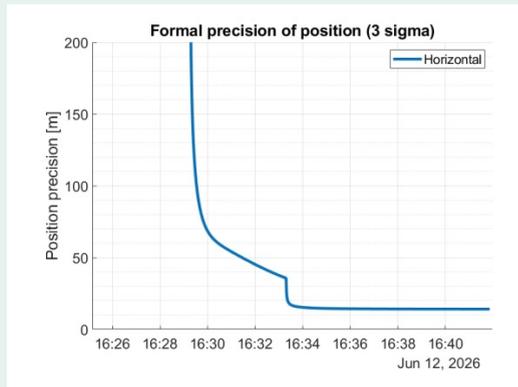
## Lunar Orbiter



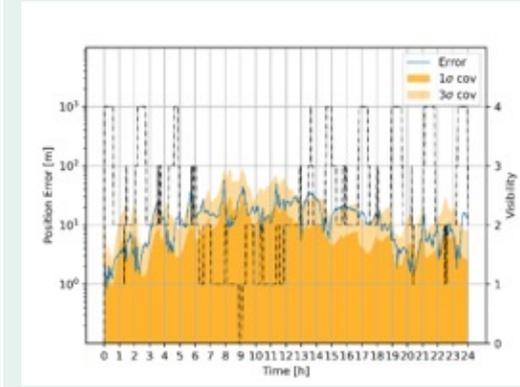
Real time  
**< 100 m (95%)**  
 LLO accuracies



**< 5 meters**



**~20 meters**



**30-60 meters**

**ALSO CONFIRMED BY INDUSTRY (AT PDR Stage)**

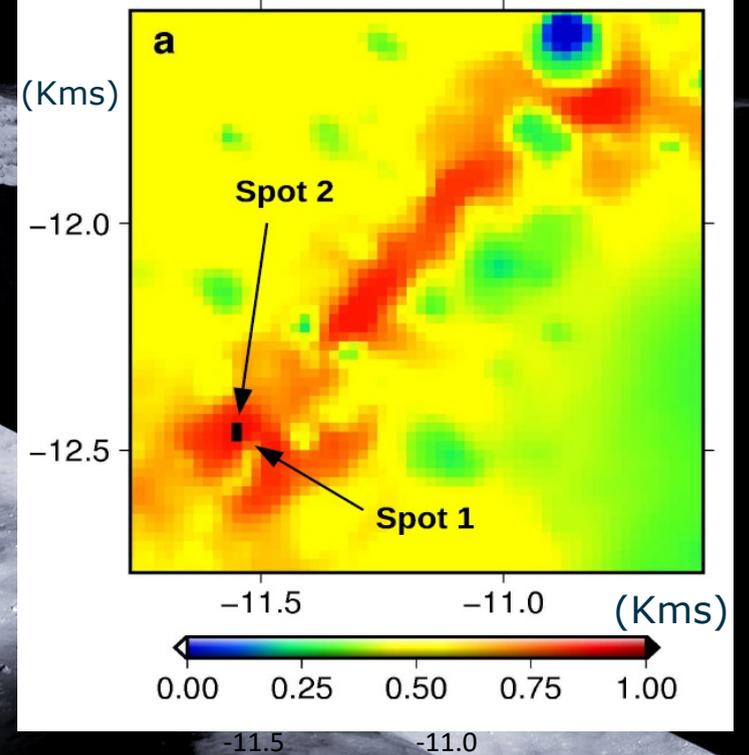
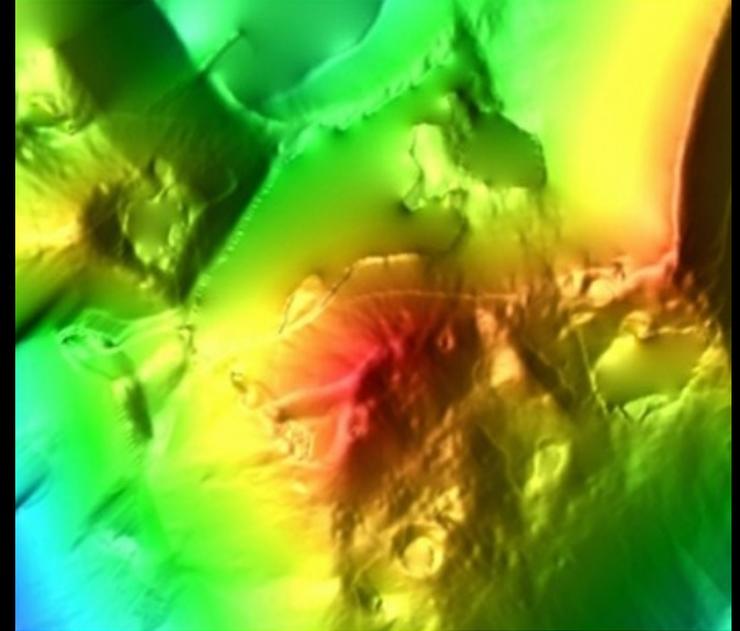
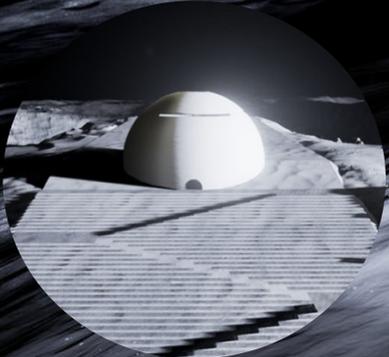


# Moonlight will make feasible landing in Peaks of Eternal Light (PEL) regions

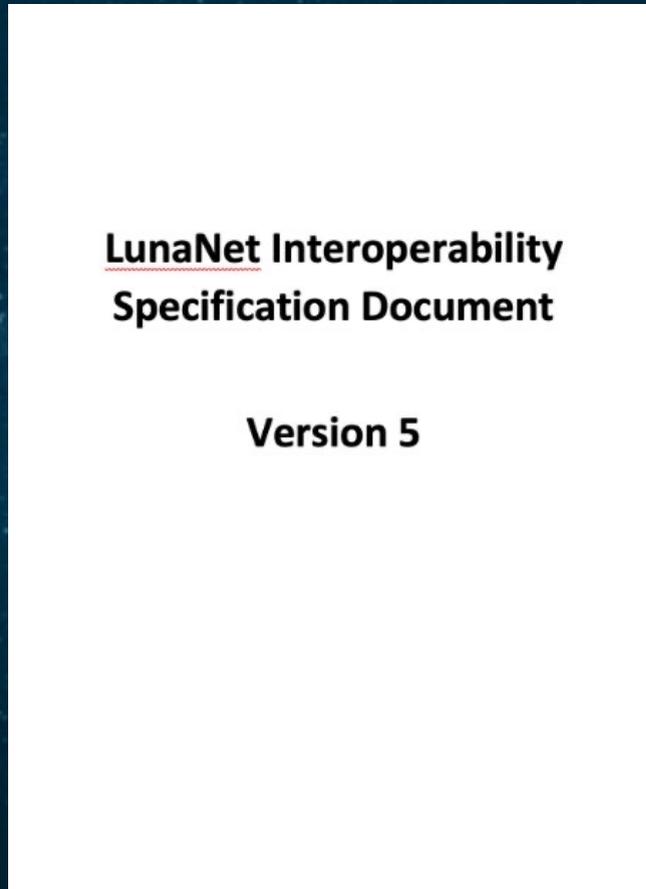
Over 20 years, the longest continuous periods in darkness are typically only 3-5 days.



Landing accuracies required  
< 50 metres



# MOONLIGHT will comply with LunaNet Interoperability Specifications



## Joint NASA, ESA and JAXA cooperation

[LunaNet V5 released in January 2025 \(working towards V6\)](#)



GNSS satellite

Test receiver at South Pole region  
On a JAXA's lander

JAXA LNSS satellite

ESA Moonlight NAV-1 satellite

NASA LCRNS satellite(s)

First-ever lunar PNT interoperability demonstration planned for 2029  
ESA / NASA / JAXA: Towards an international LANS System

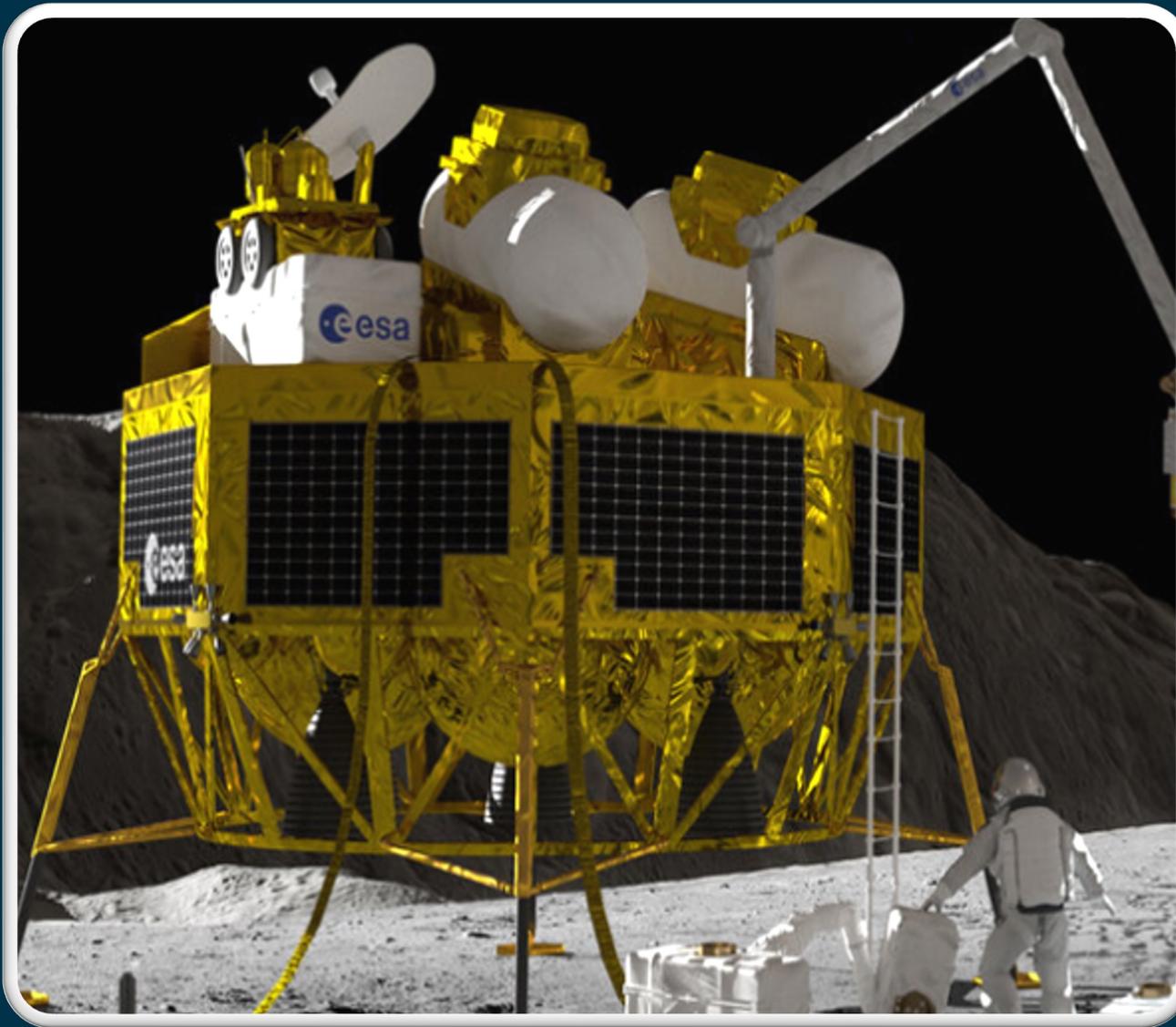
**HOW MAY WE DO BETTER (e.g. reaching ~1m range lunar surface accuracies) ON A SHORT TERM after MOONLIGHT FOC?**

## **STEP 3**

**NOVAMOON Local PNT Differential Selenodetic and time-reference Station**

**Formally approved by ESA Members States in November 2025 (CMIN25)**

**Full NovaMoon will be integrated on ESA's Argonaut-2 (2032)**



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European Space Agency

# NovaMoon towards ultimate accuracy

## 1. Reference local differential station on the lunar surface

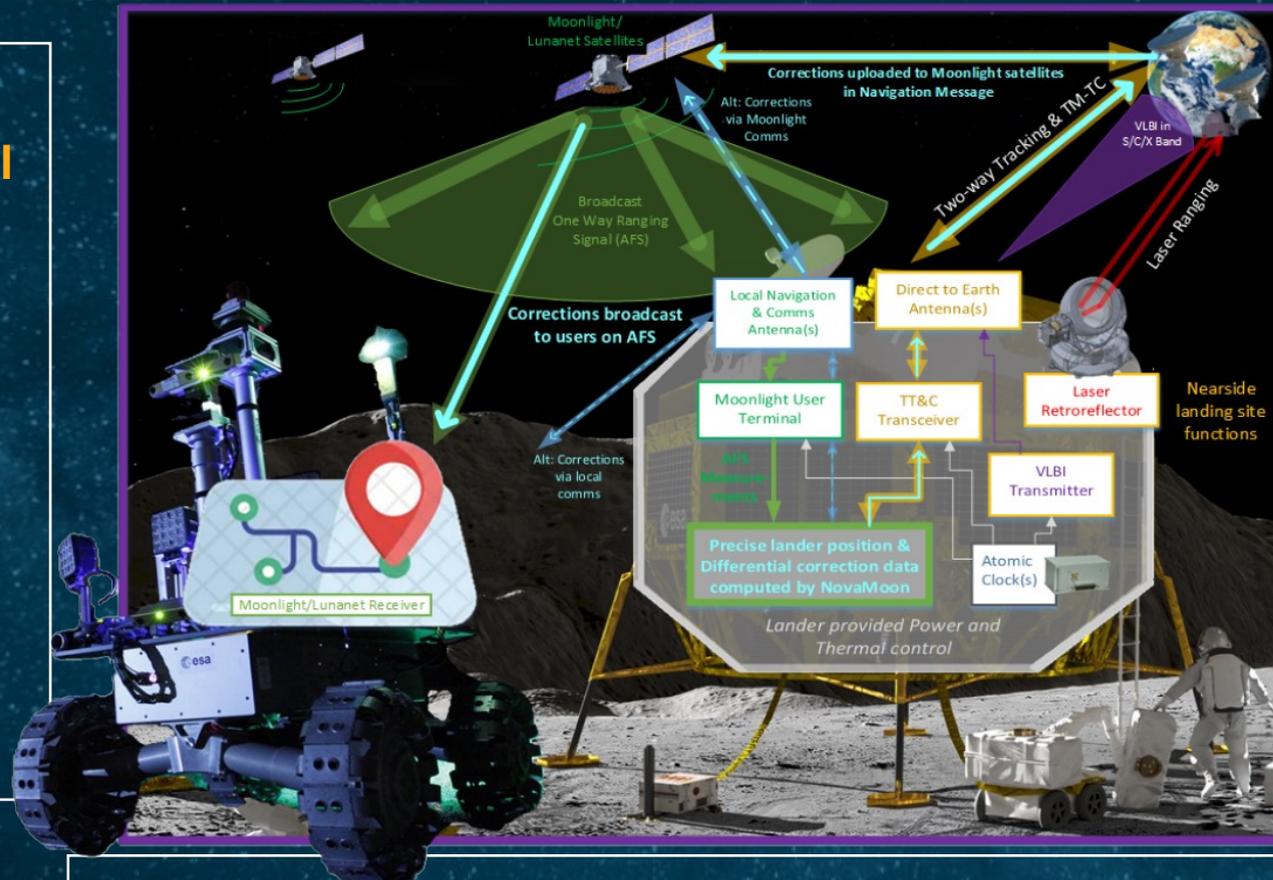
(augmenting and monitoring Moonlight and potentially other LunaNet compatible systems)

Sub-meter level accuracies over the Lunar South Pole

NovaMoon Full Payload integrated in ESA Argonaut lunar lander

5 years mission – Long-term experimentation opportunities

ESA UNCLASSIFIED – Releasable to the Public



## 3. International “Lunar Time-reference” Laboratory

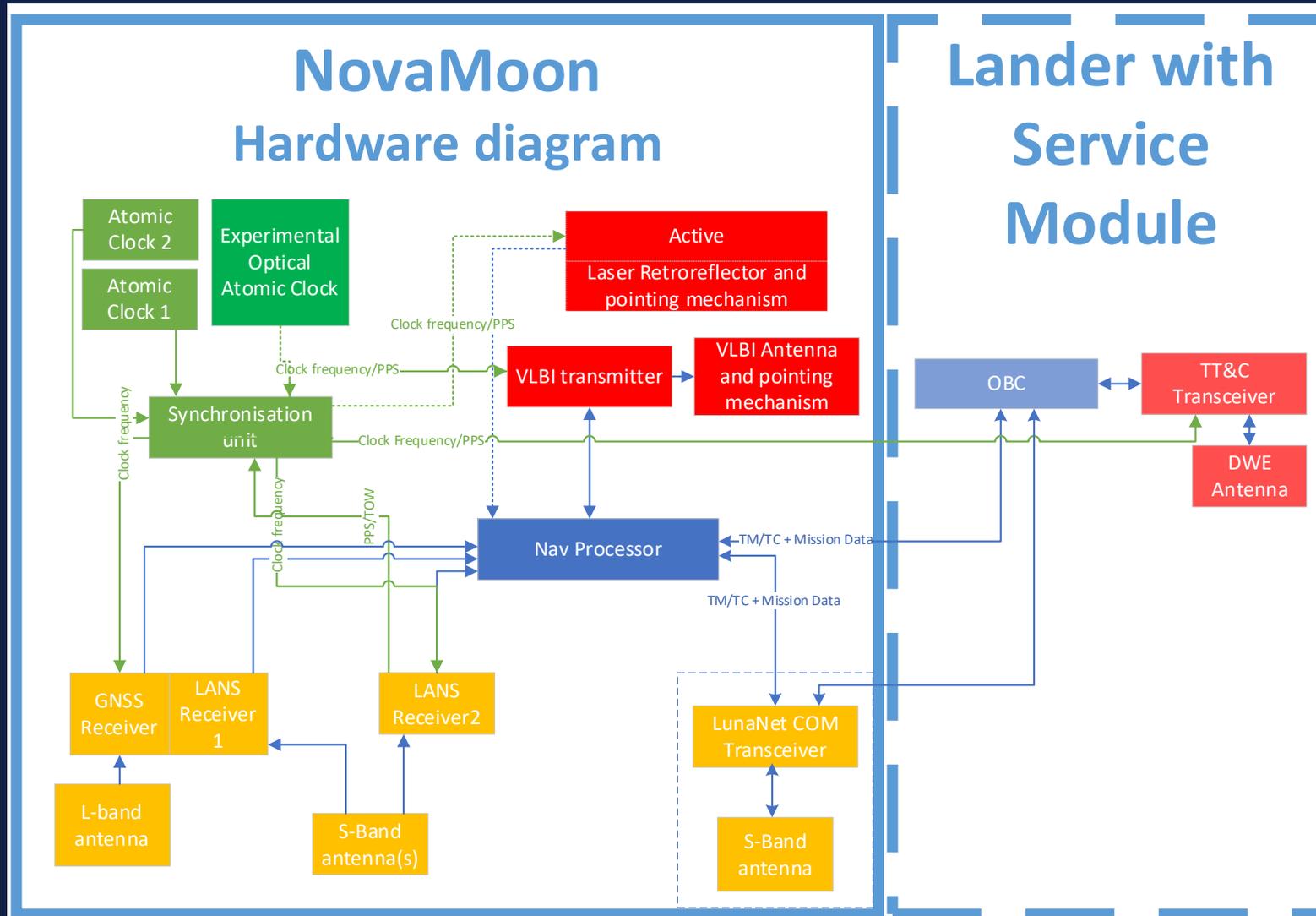
Supporting experimentation setting the basis for future lunar-base time realisations

## 2. International “geodetic” Reference Station on the lunar surface (4 collocated ranging techniques)

Towards an accurate Lunar Reference Frame realisation



# NovaMoon Complete Architecture





# NovaMoon Three incremental Builds & Three Launches !!

Step wise Approach for NovaMoon

## BUILD 1

- LunaNET + GNSS Receiver
- MiniRAFS Atomic Clock
- Advanced Laser Retroreflector

- ▶ High accuracy in static conditions
- ▶ First lunar clock on the Moon
- ▶ Moonlight / GNSS combination
- ▶ LunaNET interoperability demo



Launch 2029

## BUILD 2 adds

- Moon VLBI Transmitter
- LunaNet Rx on Lunar Rover

- ▶ First European lunar VLBI instrument
- ▶ Test IVS and VGOS experimentation
- ▶ Enhanced lunar reference frame experimentation
- ▶ High accuracy in dynamic conditions



Planned on Argonaut-1  
Launch target 2030

## BUILD 3 adds

- Additional Moonlight/LunaNET Rx
- Navigation Processor
- Additional MiniRAFS Clock
- Experimental Optical Clock
- Active Laser Ranging

- ▶ All NovaMoon services demonstrated & long-term operations (5 years !)
- ▶ All NovaMoon Scientific benefits achieved
- ▶ Moonlight High-Accuracy Service provision
- ▶ Long-term international Lunar Time, geodetic and Interoperability Platform



On Argonaut-2  
Launch target ~ 2032

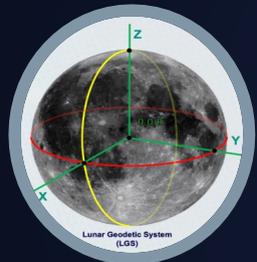
**5 Years mission !**

Moonlight NAV and LunaNET  
Monitoring & High-Accuracy Service

Lunar time realisation  
International Lunar Time  
reference laboratory



Lunar Reference  
Frame enhancement



Supporting IGS – Moon Station  
Supporting the set-up of  
Lunar-Rinex  
LUNEX standards



NovaMoon  
User Benefits



An International  
LunaNET Interoperability  
Reference Platform

GNSS and LunaNET  
satellites combined



Extracting UTC-time  
to Moon

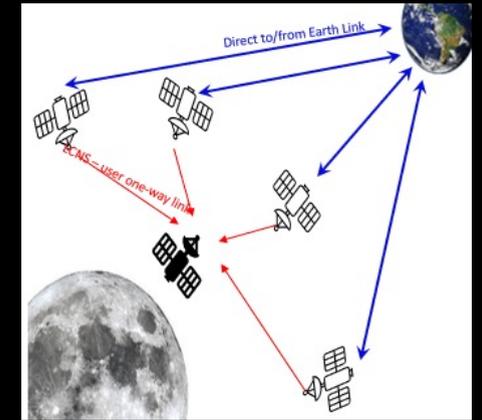
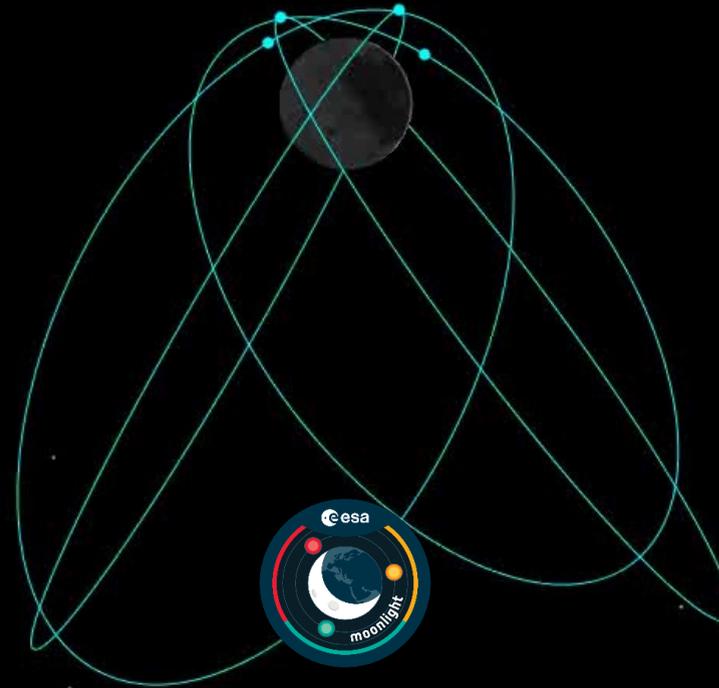
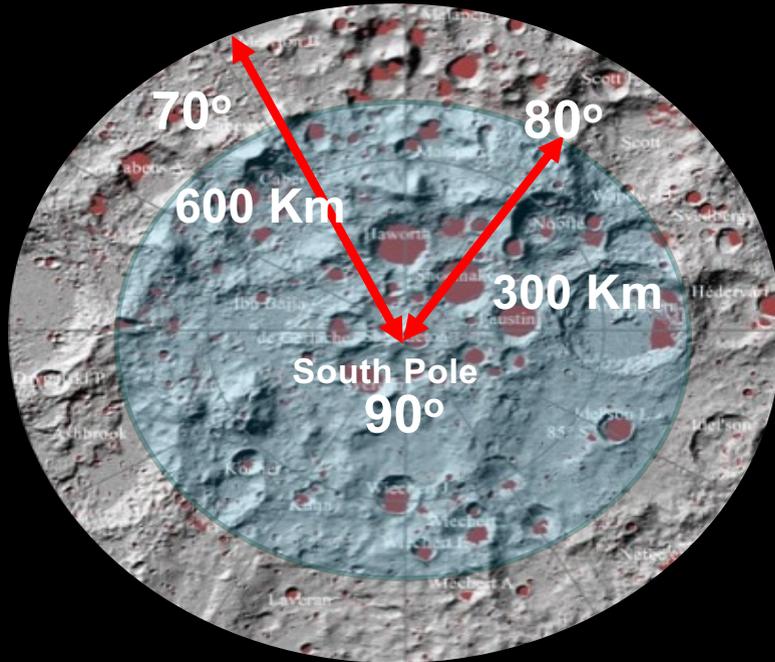
A plethora of scientific  
Applications-



Mobility tests with Argonaut rover

# Single NOVAMOON station with MOONLIGHT will provide Sub-meter accuracies over the whole South pole

ARTEMIS SERVICE AREA



ODS estimation

| LunaNet Interoperability Specifications [AD.1] |                                       |          | LSIS (this document) |        |          |
|--|---------------------------------------|----------|----------------------|--------|----------|
| MSG ID   | MSG Title                             | Category | Subframe             | Notes  | LNIS 1.0 |
| ...  |                                       |          |                      |        |          |
| MSG-G31  | Augmentation Differential Corrections | O        | SB3 and/or SB4       | Ad-hoc | yes      |



Our analysis demonstrate that a Single station will serve the whole South Pole

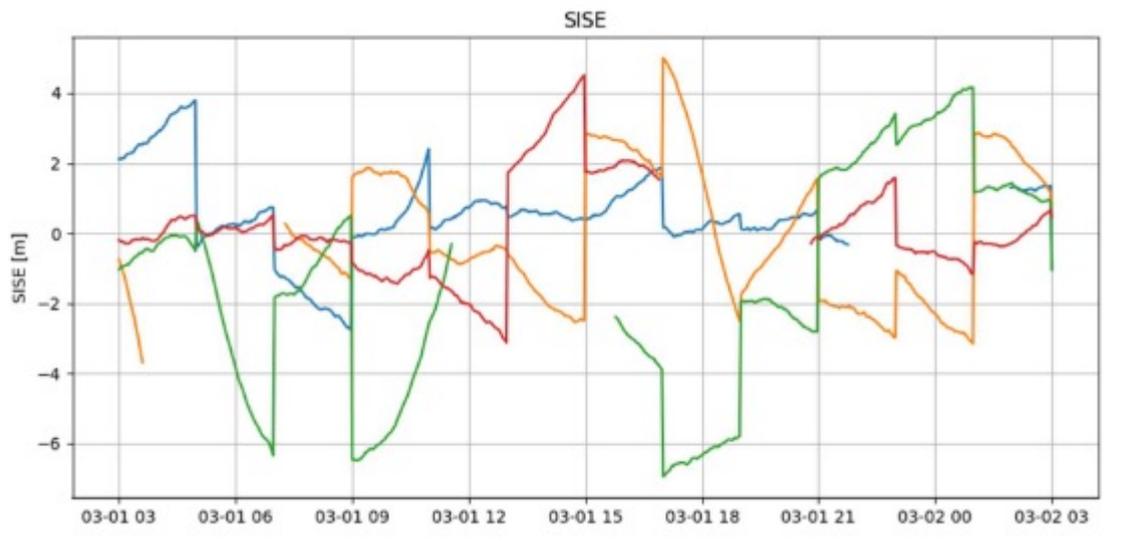
- **NO ionosphere**
- **NO troposphere**
- **Minimum Orbital projection error**
- **Lunar South pole message reception**



# Moonlight equivalent-SISE after NovaMoon corrections < 1 metre ( 1 sigma)

NovaMoon allows to compute an equivalent SISE metric defined as the residual impact of the satellite SISE on the measurements once the differential corrections are applied (5 min update)

Moonlight 4 satellites SISE (1 sigma)

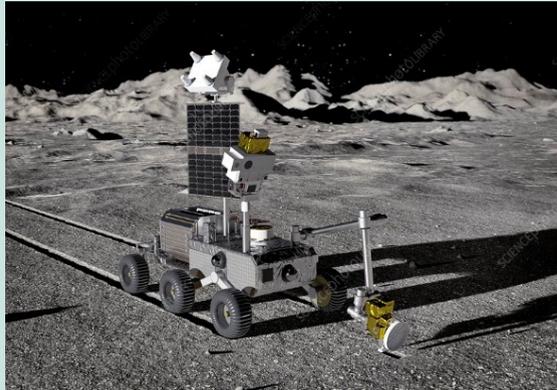


Moonlight equivalent SISE with NovaMoon (1 sigma)



# Moonlight + NovaMoon joint performances

## Surface Rover



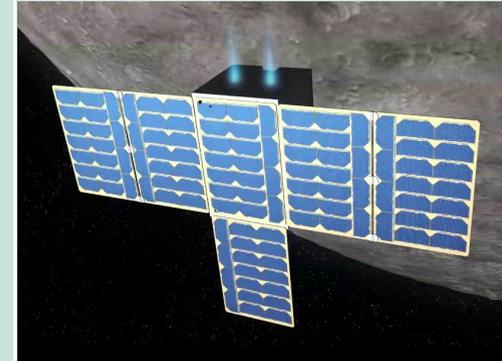
Real time  
**< 10 m (95%)**

## Lunar Lander

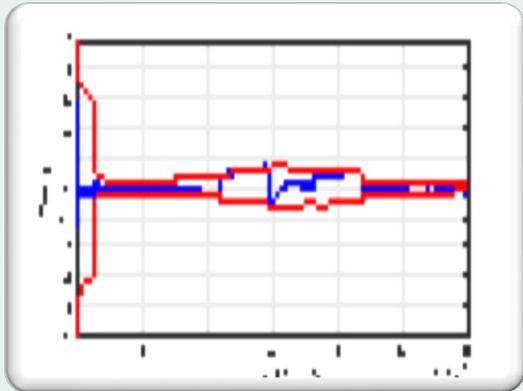


**< 50 m (95%)**  
 Landing accuracy

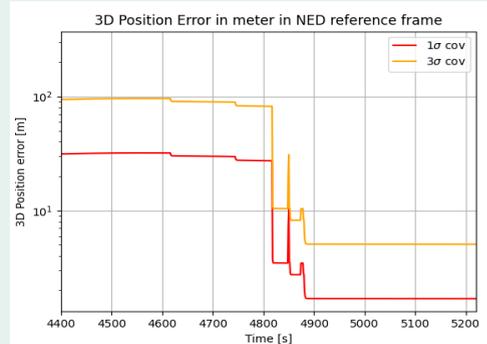
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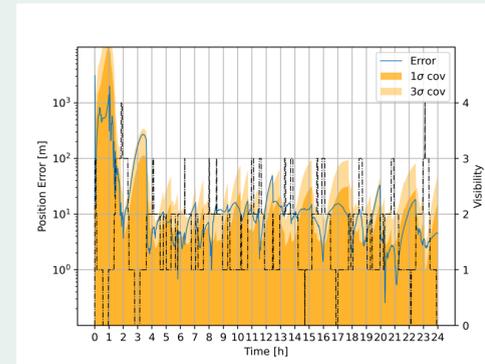
Real time  
**< 100 m (95%)**  
 LLO accuracies



  
**< 1 m**



  
**< 5 m**



  
**< 10 m**

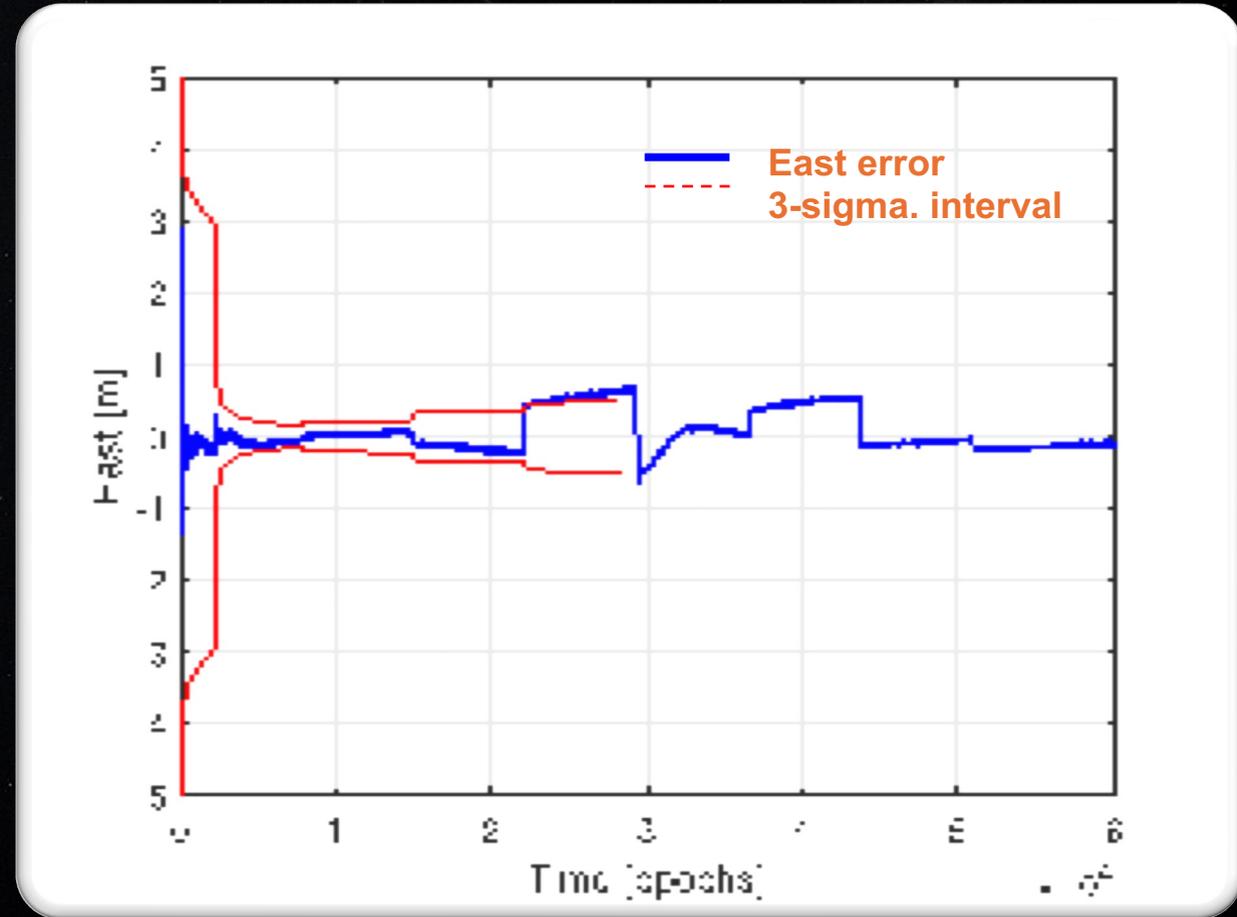
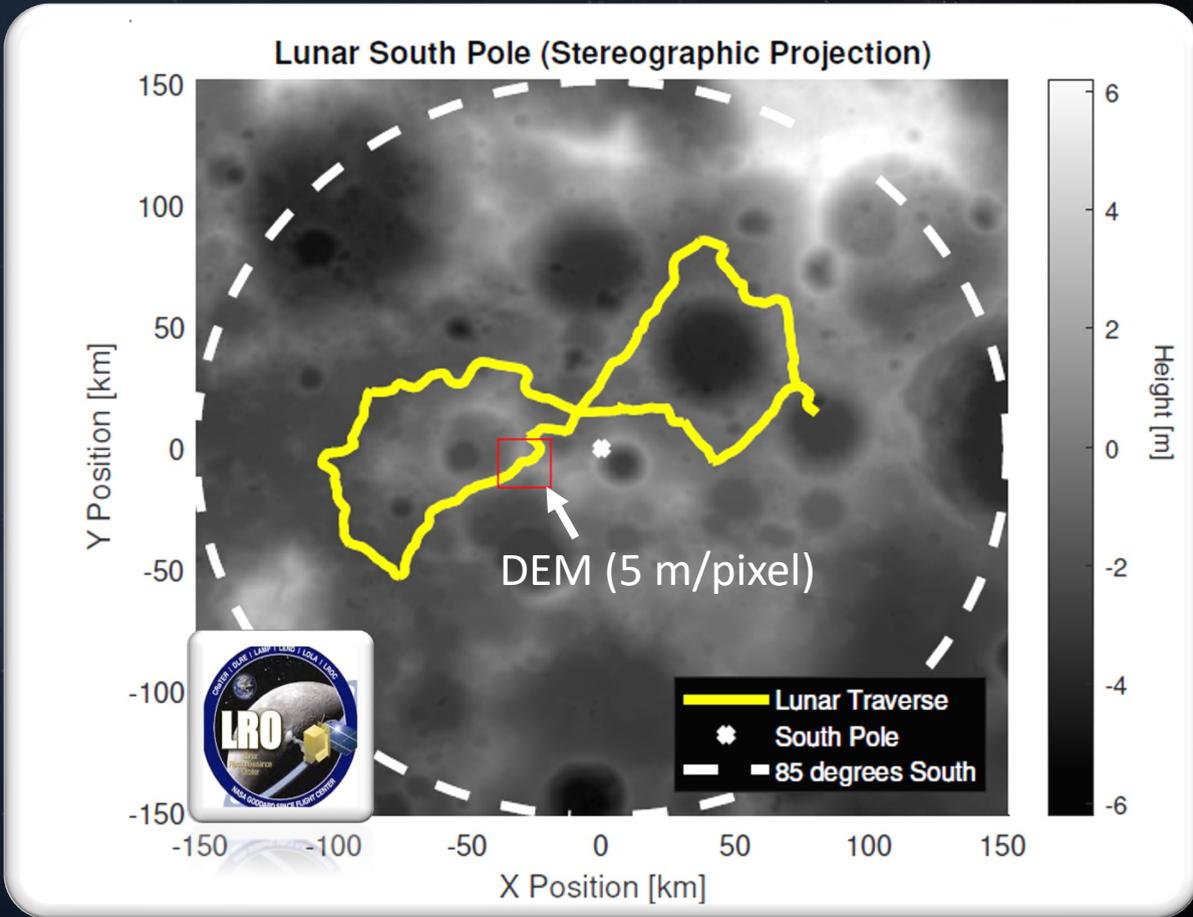
Results based on ESA and Industry Analysis (NovaMoon Phase A)





# Lunar Rover – Performance when combining Moonlight + NovaMoon

## Sub-metre Accuracy achievable in real time !

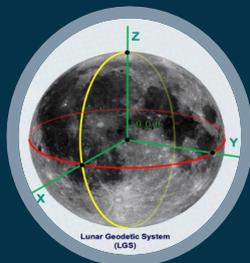


# Accuracy < 1 metre level at Lunar South pole !

Lunar time realisation  
International Lunar Time  
reference laboratory



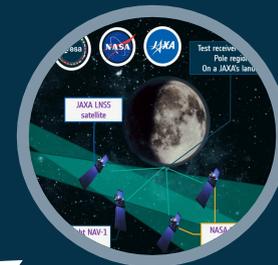
Lunar Reference  
Frame enhancement



Supporting IGS – Moon Station  
Supporting the set-up of  
Lunar-Rinex  
LUNEX standards



An International  
LunaNET Interoperability  
Reference Platform



GNSS and LunaNET  
satellites combined



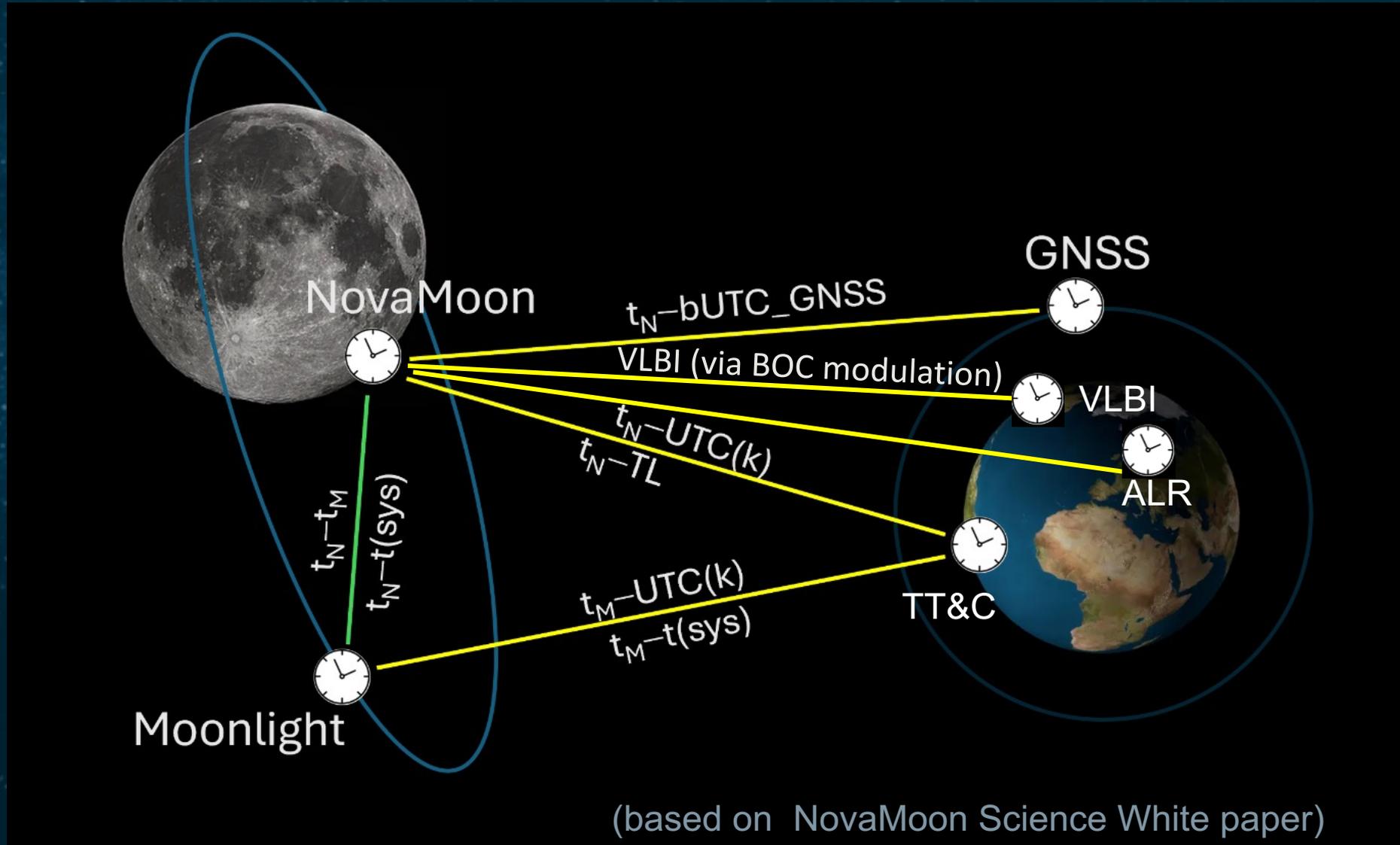
Extracting UTC-time  
to Moon

A plethora of scientific  
Applications-



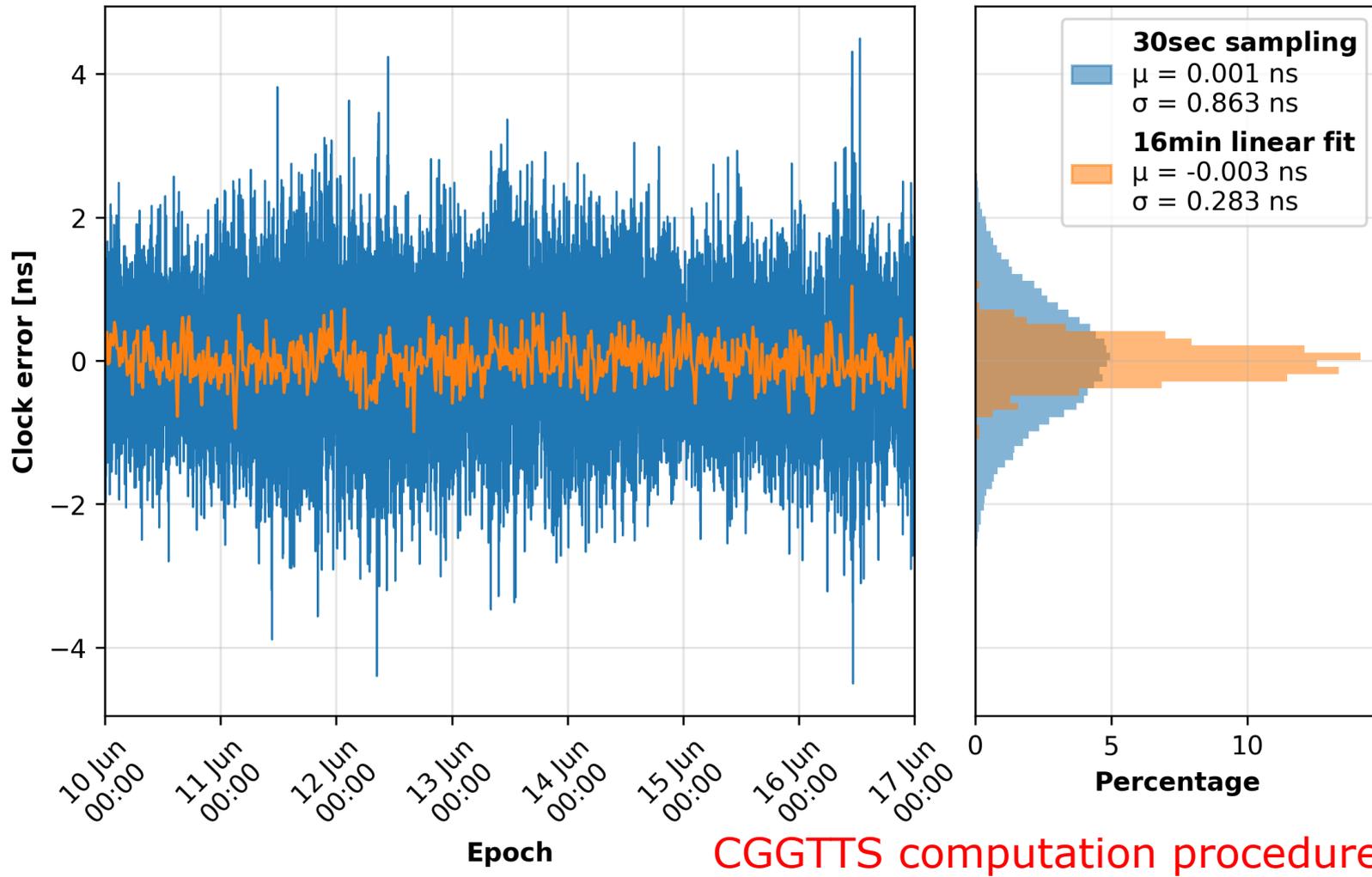
Mobility tests with Argonaut rover

# NovaMoon: A unique Lunar Time Laboratory



# NovaMoon: Example of GNSS UTC Time transfer

Lunar lander at 90°S, 0°E with **MiniRAFS** using E5a+L5 & **HGA**



**GNSS time transfer shows to be a good solution to access UTC on NovaMoon in real-time.**

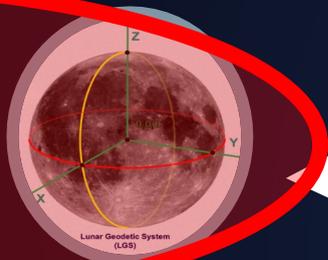
**Additional available Time-Transfer method allow for higher accuracy**

**This accurate time-transfer could support the characterization of relativistic offsets between Moon- and Earth-based clocks**

Lunar time realisation  
International Lunar Time  
reference laboratory

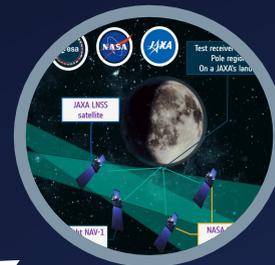


Lunar Reference  
Frame enhancement



**NovaMoon  
User Benefits**

An International  
LunaNET Interoperability  
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GNSS and LunaNET  
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Extracting UTC-time  
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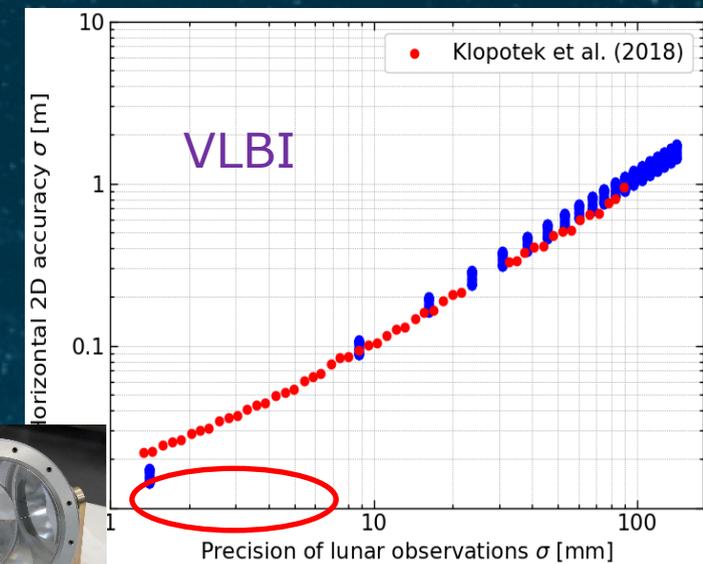
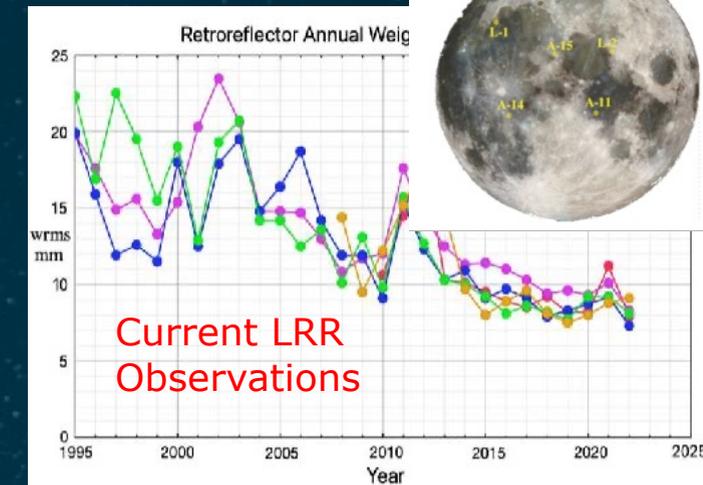
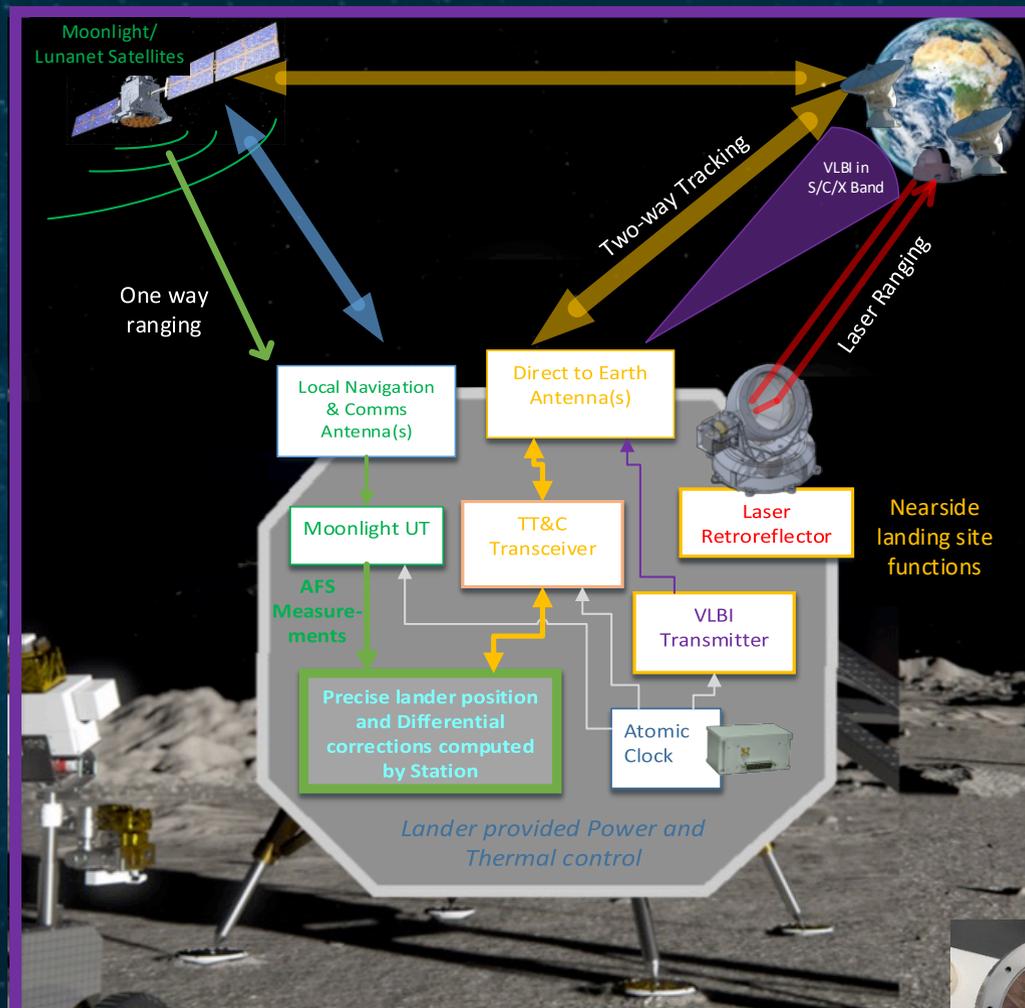
# Lunar Base Station Localisation

cm-level (< 10 cm) position determination on the lunar surface !

For the first time 4 co-located geodetic techniques on-board:

- Augmented PNT LunaNET
- VLBI transmission
- Laser Ranging Retroreflector
- Two-way Direct To Earth

**3D localisation accuracies down to cm-level are at reach !**



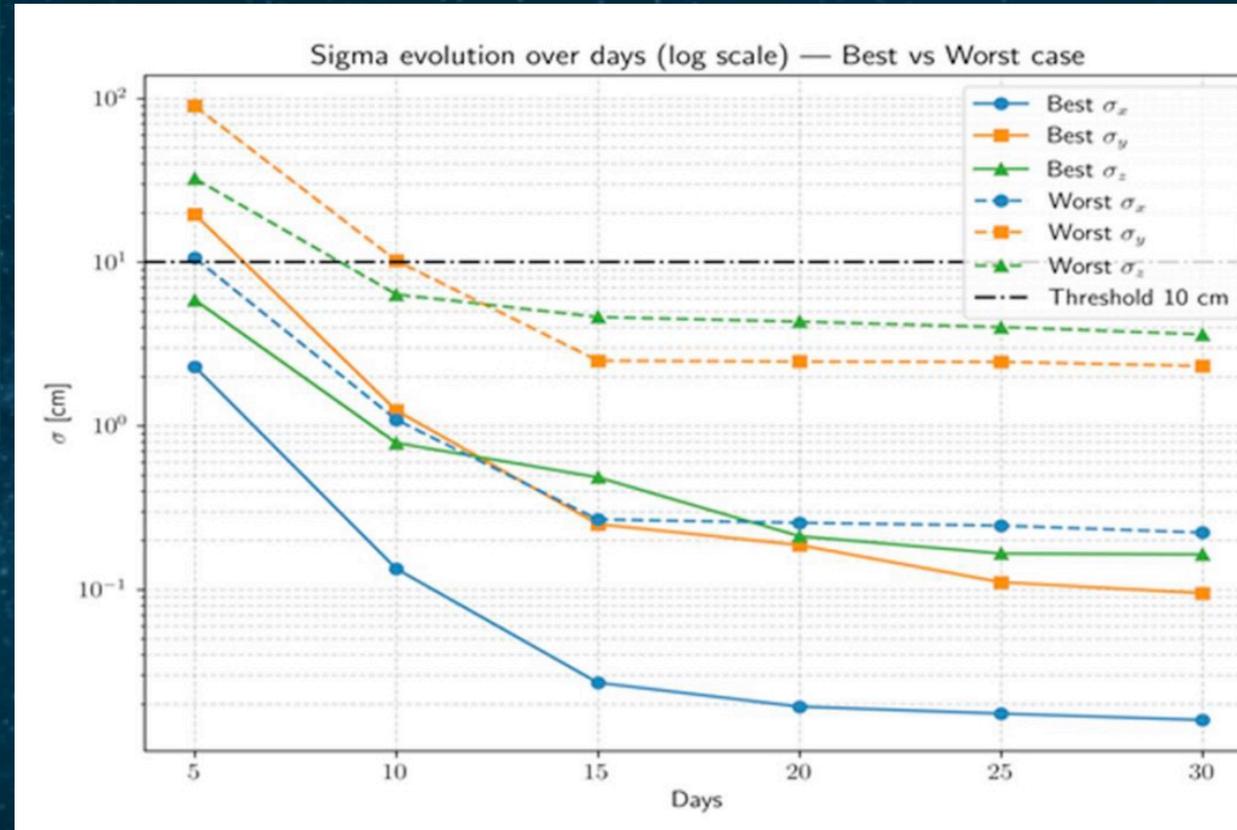
# Lunar Base Station Localisation NovaMoon Station cm-level (< 10 cm) position determination on the lunar surface !

## Reference Position 1: 85S 3E (Example)

For the first time 4 co-located geodetic techniques on-board:

- Augmented PNT LunaNET
- VLBI transmission
- Laser Ranging Retroreflector
- Two-way Direct To Earth

**3D localisation accuracies down to cm-level are at reach !**



Worst case scenario (30 days) [cm]

|            |      |
|------------|------|
| $\sigma_x$ | 0.22 |
| $\sigma_y$ | 2.32 |
| $\sigma_z$ | 3.62 |

Best case scenario (30 days) [mm]

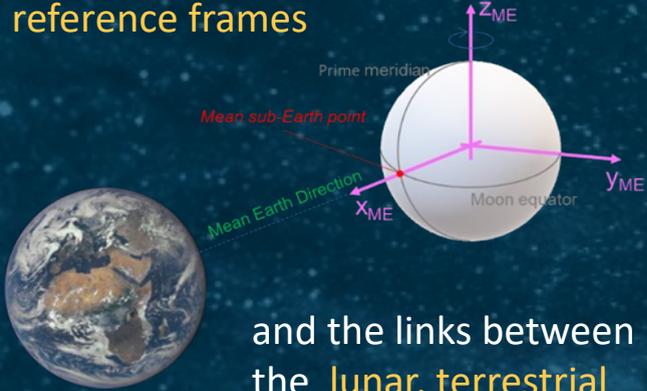
|            |      |
|------------|------|
| $\sigma_x$ | 0.16 |
| $\sigma_y$ | 0.95 |
| $\sigma_z$ | 1.64 |

(Source CRAS, ESA NovaMoon Phase A Study)

~10 days of observation allow to reach a localisation accuracy < 10 cm  
Could become a reference standard for Selenodetic Lunar-base stations

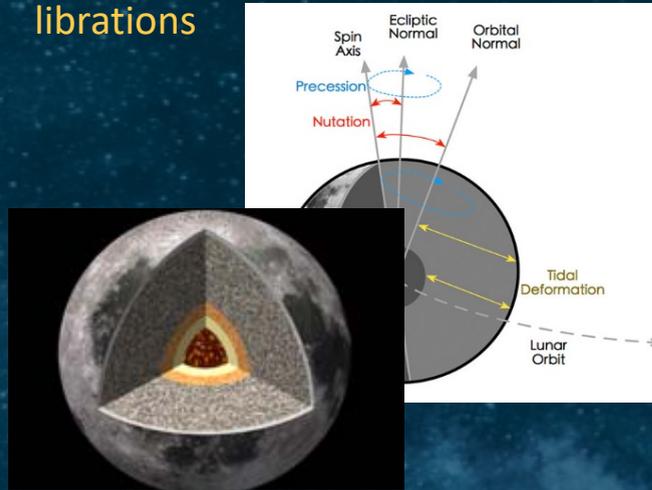
# NovaMoon VLBI a major asset to enhance Lunar Reference Frame and Lunar Geophysics knowledge

Enhancing the accuracy of Lunar reference frames



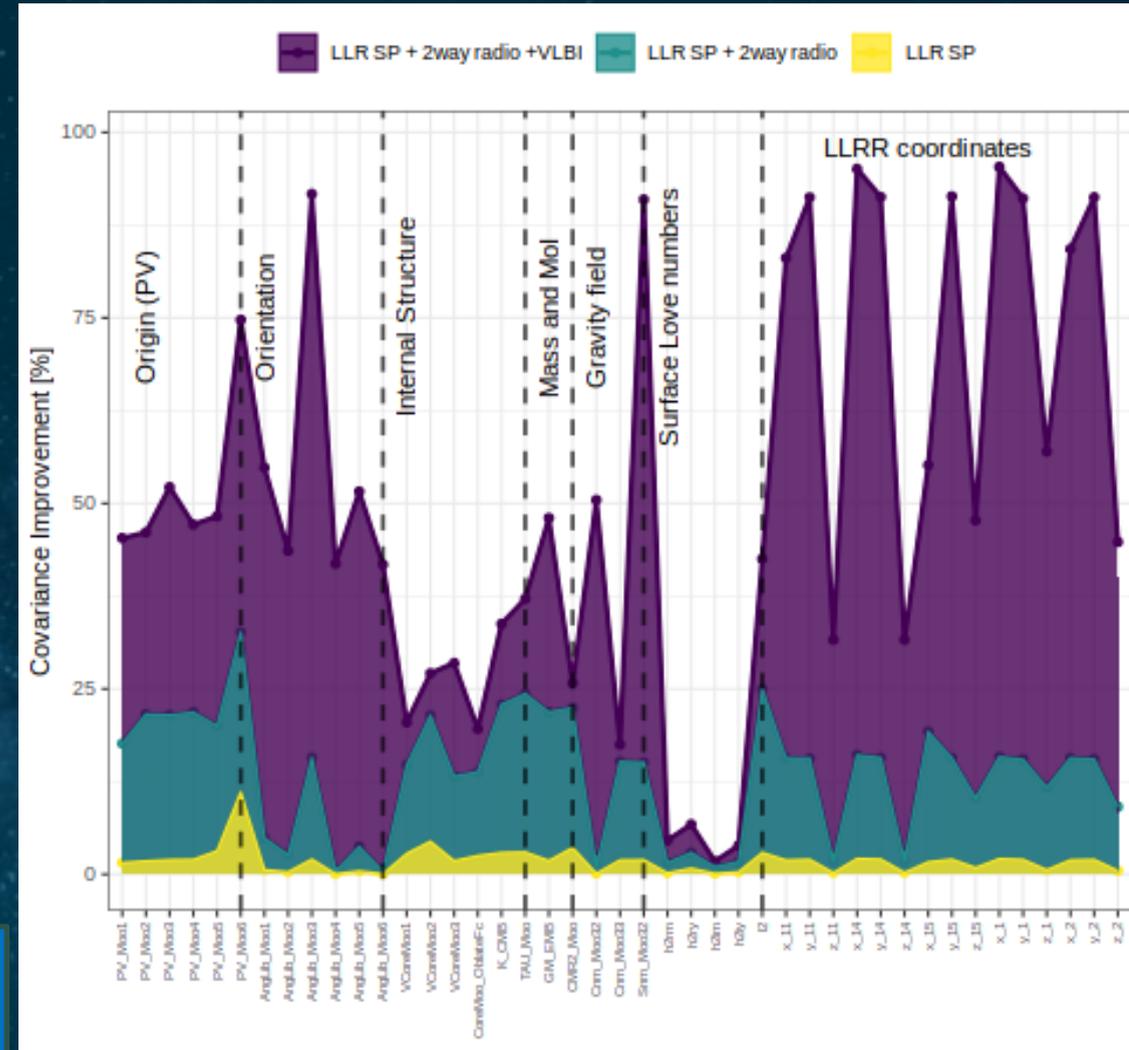
and the links between the lunar, terrestrial and celestial frames

Supporting measurements of the Moon tidal deformations and Moon librations



Improving the knowledge of the Moon Position/Velocity, Orientation, Internal structure, lunar gravity field, Mol, Surface love Numbers, Moon core dynamics,

Relative improvements (%) in the Lunar Reference Frame (LRF) through covariance matrix analysis of 40 years of LLR data.

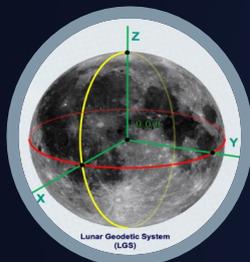


(Credits: A. Fienga, NovaMoon Science White Paper)

Lunar time realisation  
International Lunar Time  
reference laboratory



Lunar Reference  
Frame enhancement



Supporting IGS – Moon Station  
Supporting the set-up of  
Lunar-Rinex  
LUNEX standards

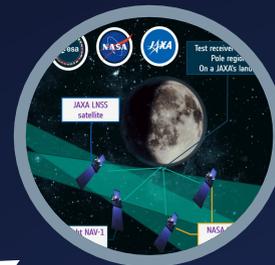


**NovaMoon  
User Benefits**



Mobility tests with Argonaut rover

An International  
LunaNET Interoperability  
Reference Platform



GNSS and LunaNET  
satellites combined



Extracting UTC-time  
to Moon



# NovaMoon at the service of Science:

## NovaMoon Science White paper produced

The NovaMoon Science White Paper presents a very comprehensive set of **primary and secondary scientific potential user benefits (objectives)** identified for the NovaMoon mission.

This document is **supported by 104 reputed Scientists** (84 having directly contributed to the paper), **representing 40 different European Research institutions**, covering a large variety of complementary fields.

This document has been submitted for publication to **Space Science Reviews** (Springer - Dec 2025)

**Preprint available from today at arXiv repository**  
<https://doi.org/10.48550/arXiv.2602.08432>



# ESA strong will to support Lunar PNT users benefits & to promote International collaboration 8 times on cislunar space supporting PNT between 2027-2032 !!



Lunar Pathfinder

LPF (2027)



Moonlight NAV-1 2029  
Moonlight NAV 2, 3 & 4 2031



NOVAMOON

NovaMoon Build#1 2029  
NovaMoon Build#2 2030  
NovaMoon Build#3 2032